THE VEHICLE THE SATELLITE

PROTON HISTORY

■ Lead designer was Vladimir Chelomei, who designed it with the intention of creating both a powerful rocket for military payloads and a high-performance ICBM. The program was changed, and the rocket was developed exclusively for launching spacecraft.

■ First named UR-500, but adopted the name "Proton," which also was the name of the first three payloads launched.

Proton launched Russian interplanetary missions to the Moon, Venus, Mars, and Halley's Comet.

■ Proton launched the Salyut space stations, the Mir core segment and both the Zarya (Dawn) and Zvezda (Star) modules for today's International Space Station.

■ First commercial Proton launch — 9 April 1996.

■ First commercial Proton M Breeze M launch
— 30 December 2002

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TOTAL HEIGHT 58.2 m (191 ft)

GROSS LIFT-OFF
WEIGHT
705,000 kg
(1,554,000 lb)

PROPELLANT UDMH and NTO

INITIAL LAUNCH 16 July 1965 Proton-1 Spacecraft

PAYLOAD FAIRINGS
There are multiple payload
fairing designs presently
qualified for flight, including
standard commercial payload
fairings developed specifically to
meet the needs of our customers.

Breeze M Upper Stage
The Breeze M is powered by one pump-fed
gimbaled main engine that develops thrust of
20 kN (4,500 lbf). It is composed of a central core and
an auxilliary propellant tank which is jettisoned in flight
following depletion. The Breeze M control system includes an
on-board computer, a three-axis gyro stabilized platform, and a
navigation system. The quantity of propellant carried is dependent
on specific mission requirements and is varied to maximize mission
performance.

PROTON BOOSTER

The Proton booster is 4.1 m (13.5 ft) in diameter along its second and third stages, with a first stage diameter of 7.4 m (24.3 ft). Overall height of the three stages of the Proton booster is 42.3 m (138.8 ft).

THIRD STAGE

Powered by one RD-0213 engine, this stage develops thrust of 583 kN (131,000 lbf), and a four-nozzle vernier engine that produces thrust of 31 kN (7,000 lbf). Guidance, navigation, and control of the Proton M during operation of the first three stages is carried out by a triple redundant closed-loop digital avionics system mounted in the Proton's third stage.

SECOND STAGE

Of conventional cylindrical design, this stage is powered by three RD-0210 engines plus one RD-0211 engine and develops a vacuum thrust of 2.4 MN (540,000 lbf).

FIRST STAC

The first stage consists of a central tank containing the oxidizer surrounded by six outboard fuel tanks. Each fuel tank also carries one of the six RD-276 engines that provide first stage power. Total first stage vacuum-rated level thrust is 11.0 MN (2,500,000 lbf).

The Proton and the Breeze M are built by Khrunichev State Research and Production Space Center.



SATELLITE OPERATOR

Yahsat www.yahsat.ae

SATELLITE MANUFACTURER

Astrium www.astrium.eads.net

Thales Alenia Space www.thalesaleniaspace.com

PLATFORM Eurostar E3000

 $\mathsf{S}\mathsf{eparated}\;\mathsf{M}\mathsf{ass}$

< 6,100 kg

SATELLITE MISSION LIFETIME
15 Years

SATELLITE MISSION

Second Yahsat satellite to be launched in one year, Y1B will deliver communications in Ka band for both commercial and governmental users. It will provide high-data rate internet services for public and private users in the Middle East, Africa and Southwest Asia above the limitations of existing terrestrial and satellite systems. Y1B commercial communication payload uses state of the art Ka-band multi spot-beam technology, achieving cost-effective bandwidth supply through 61 narrow spot beams. The governmental mission will bring a substantial increase in the total capacity available for the UAE secured communications over Middle East in complement to the Y1A satellite.



Mission Overview



Experience ILS: Achieve Your Mission

QUALITY | PERFORMANCE | EXPERIENCE | DEDICATION



www.ilslaunch.com

Y₁B

- 1st Yahsat Satellite Launched on ILS Proton
- 3rd ILS Proton Launch in 2012
- 14th Eurostar Satellite Launched on ILS Proton
- 72nd ILS Proton Launch Overall

THE MISSION

Ignition

03:28:09

Shutdown

03:39:23

10

APT Jettison

03:40:13

Ignition

03:41:40

Shutdown

01:25:16



MISSION DESCRIPTION

The Proton M launch vehicle, utilizing a 5-burn Breeze M mission design, will lift-off from Pad 39 at Baikonur Cosmodrome, Kazakhstan, with the Y1B satellite on board. The first three stages of the Proton will use a standard ascent profile to place the orbital unit (Breeze M upper stage and the Y1B satellite) into a sub-orbital trajectory. From this point in the mission, the Breeze M will perform planned mission maneuvers to advance the orbital unit first to a circular parking orbit, then to an intermediate orbit, followed by a transfer orbit, and finally to a geostationary transfer orbit. Separation of the Y1B satellite is scheduled to occur approximately 9 hours, 12 minutes after liftoff.

GROUND TRACK

Shutdown

03:47:49

m 🕡

Ignition

08:53:20

Shutdown

08:59:24

Spacecraft

Separation

09:12:00



